



DEVELOPMENT OF A NEW COLD-FORMED TRUSS SYSTEM

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BACKGROUND

R&D project with industrial background

Aim

- Development of a truss system and design method
- Verification/validation of the design method

Main characteristics of the system

- Span: 12...24 meter
- Using only cold-formed C-sections
- Flexible system allowing free design
- Out-of-box solutions

STRUCTURAL ARRANGEMENT

Structural elements

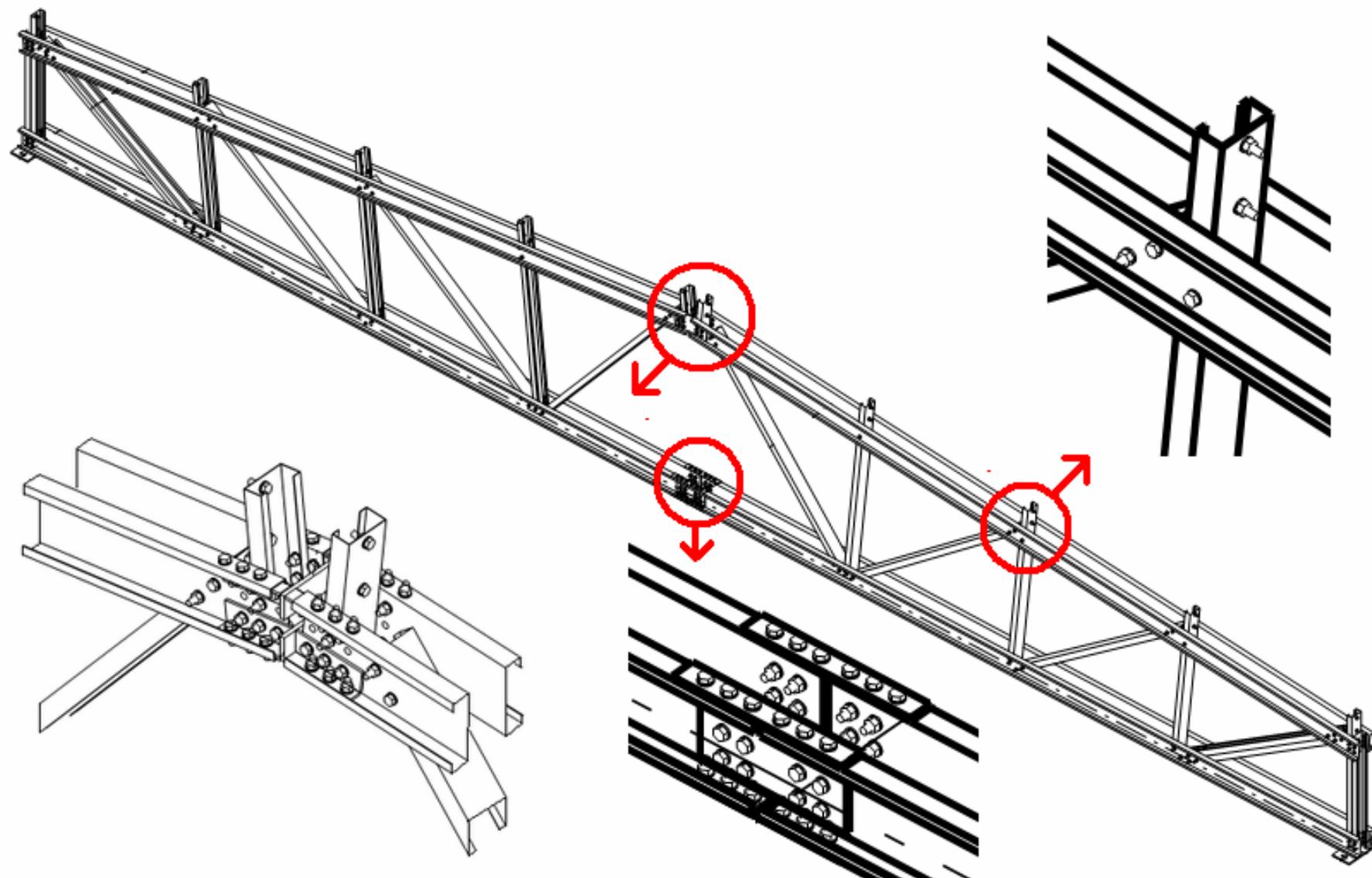
- Chord: two C-sections in back-to-back arrangement
- Bracing: single C-sections, doubled at the supports

Structural joints

- Eccentric bolted connections



STRUCTURAL ARRANGEMENT



PROBLEM STATEMENT

CHORDS
+
BRACING



JOINT

Stability issues ✓

– EC-based design

Similar problems solved

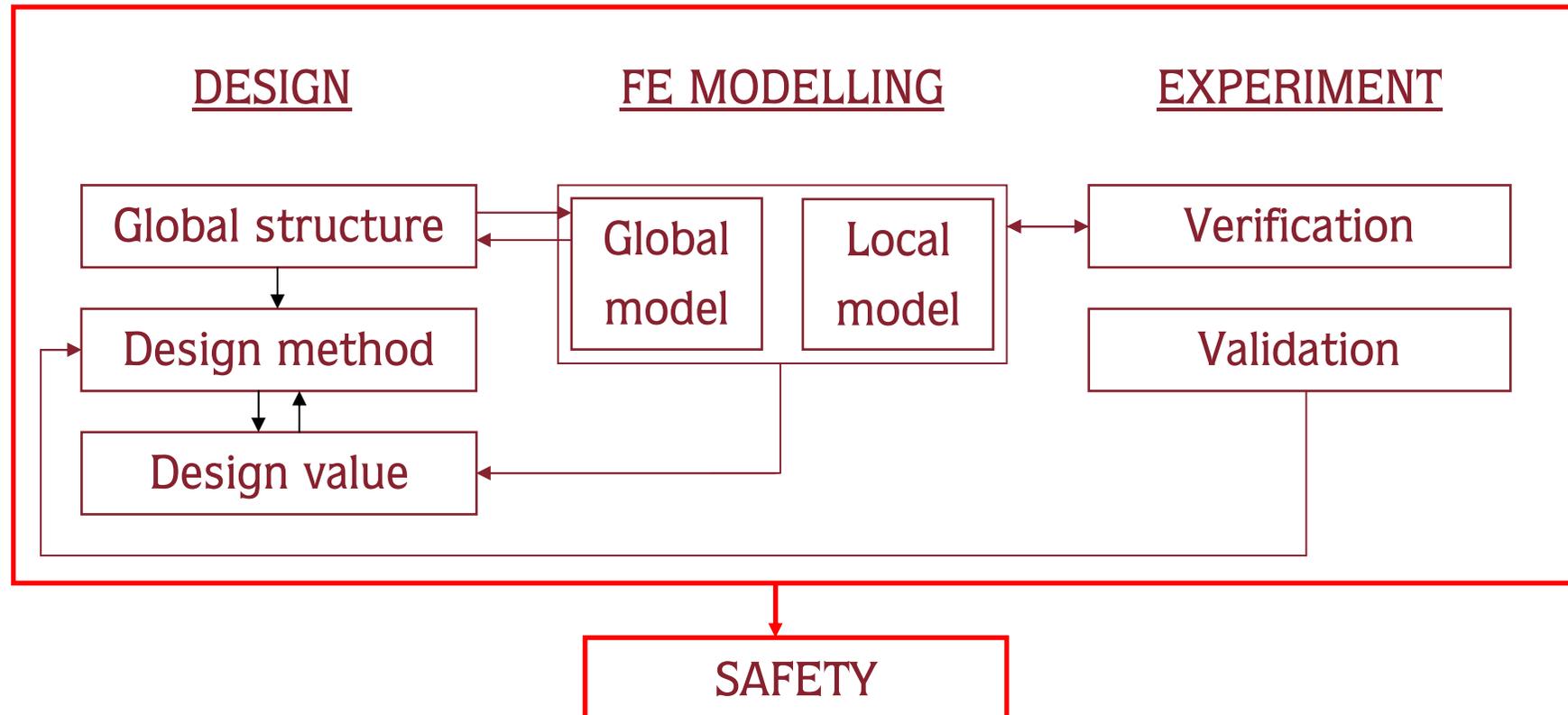
Eccentricity ?

– in-plane, out-of-plane

Load-bearing capacity, local stability behaviour

Rigidity, interaction with members

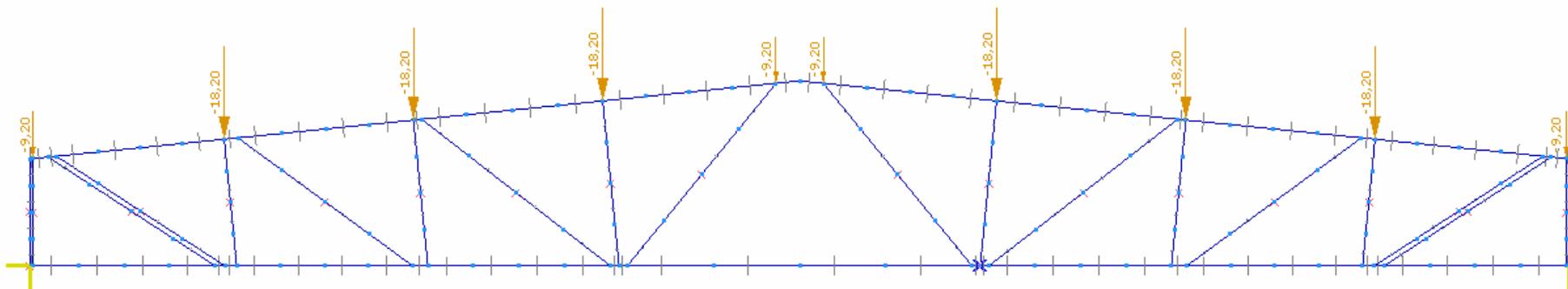
SOLUTION STRATEGY



FE MODELLING – BEAM MODEL

Global model – for design

- 2-D beam model taking into account only in-plane eccentricities
- Used to determine the internal forces needed for design



DESIGN METHOD – MEMBERS

BRACE MEMBERS

- Dominant mode: Compression and bending or interaction of flexural buckling and bending about the weak axis

CHORD MEMBERS

- Dominant mode: Interaction of flexural buckling and biaxial bending

$$\frac{N_{Ed}}{\chi_z \cdot A_{eff} \cdot f_{yb} / \gamma_{M1}} + \frac{\kappa_y \cdot M_{y,Ed}}{W_{eff,y} \cdot f_{yb} / \gamma_{M1}} + \frac{\kappa_z \cdot N_{Ed} \cdot e_{N,y}}{W_{eff,z} \cdot f_{yb} / \gamma_{M1}} \leq 1$$

$$45.8\% + 7.8\% + 46.4\% = 100\%$$



DESIGN METHOD – JOINTS

BOLTED CONNECTION

- Shear failure
- Dominant mode: bearing failure

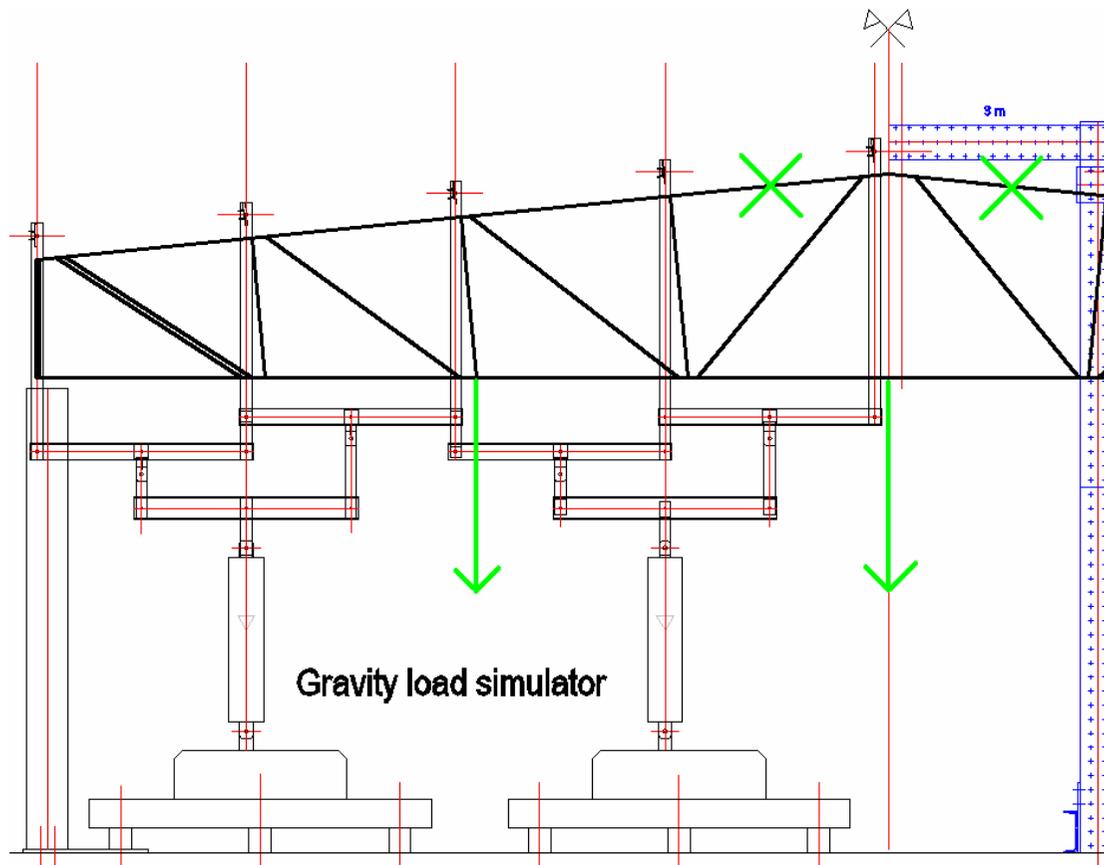
NO JOINT FAILURE MODES CONSIDERED



LABORATORY TESTING - SETUP

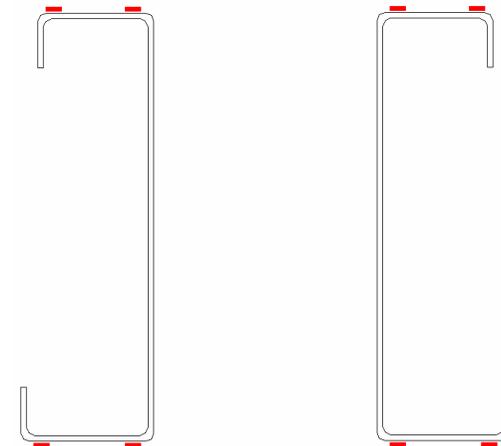


LABORATORY TESTING - SETUP



MEASUREMENT

Load via oil pressure
Strains – strain gage
Displacements



LABORATORY TESTING – FAILURE MODES

Test 1

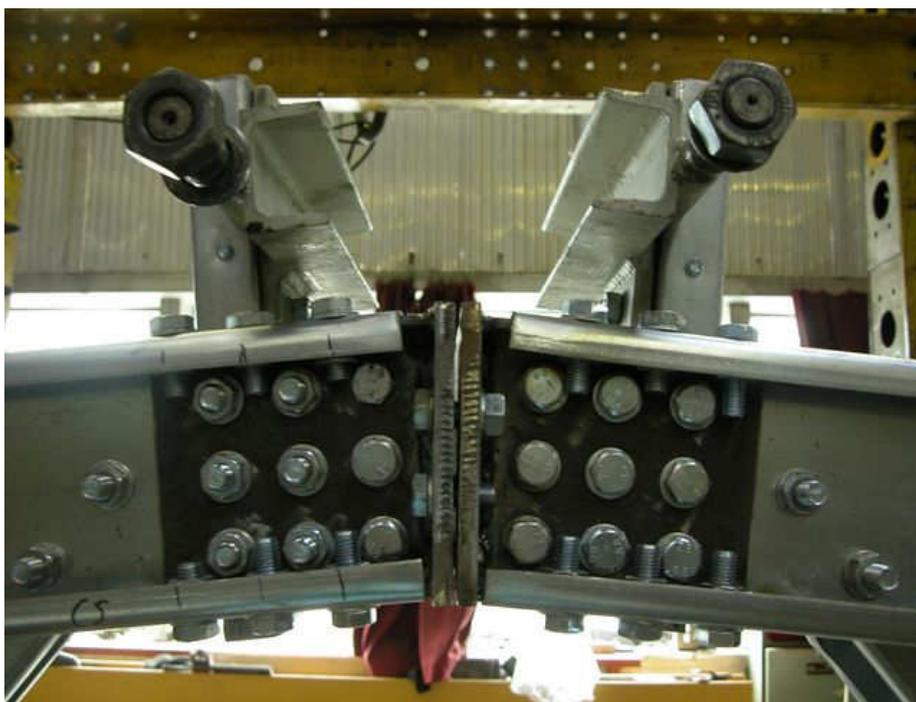


failure in the upper chord, interaction of bending (weak axis) and flexural buckling

Load: 28,5 kN/jack

LABORATORY TESTING – FAILURE MODES

Test 2



failure of the peak
joint

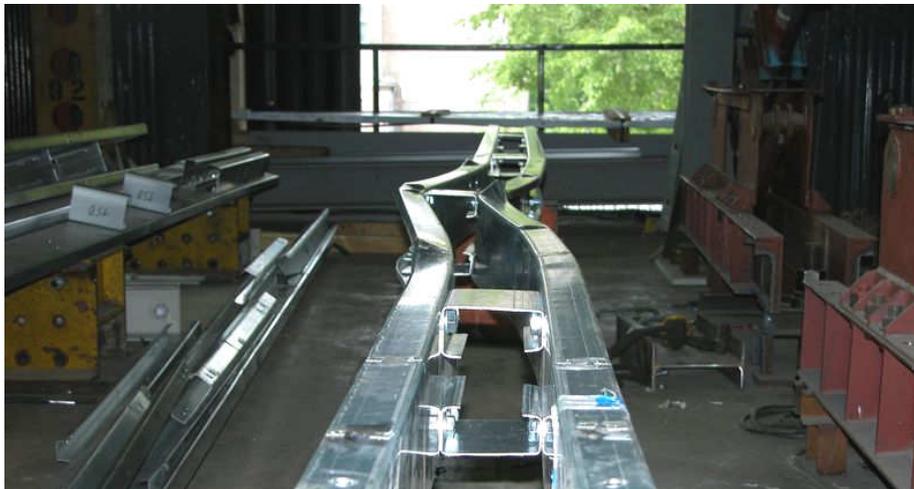
Load: 35,5 kN/jack



New peak joint
arrangement

LABORATORY TESTING – FAILURE MODES

Test 3

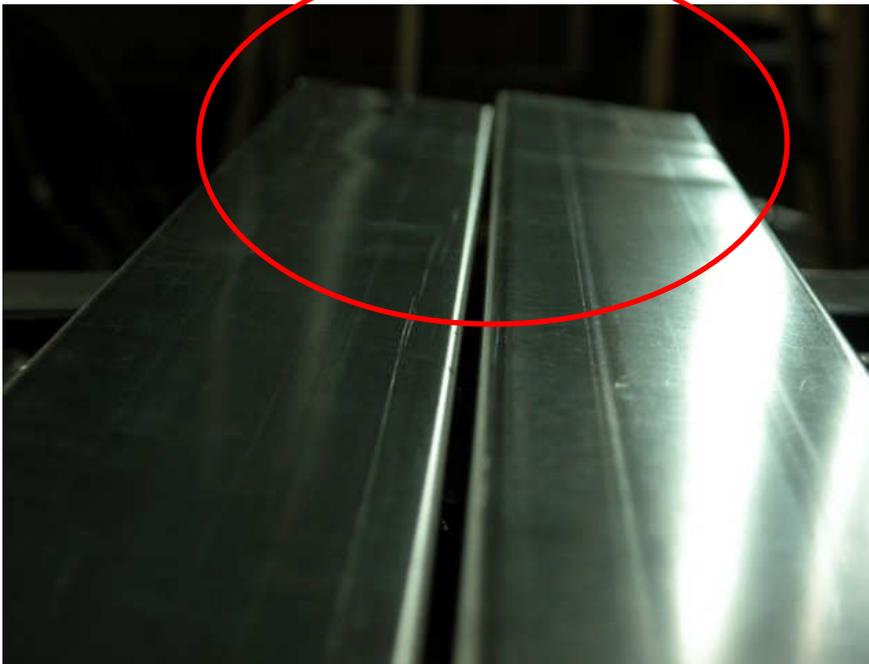


failure of the upper
chord (built-up section)

Load: 36,4 kN/jack

LABORATORY TESTING – FAILURE MODES

Test 4



failure of a compression
brace member;
interaction of
compression and bending

Load: 37,4 kN/jack

LABORATORY TESTING – FAILURE MODES

Test 5 – first failure



failure in the lower chord joint nearest to the support; interaction of shear and tension

Load: 37 kN/jack

LABORATORY TESTING – FAILURE MODES

Test 5 – final failure



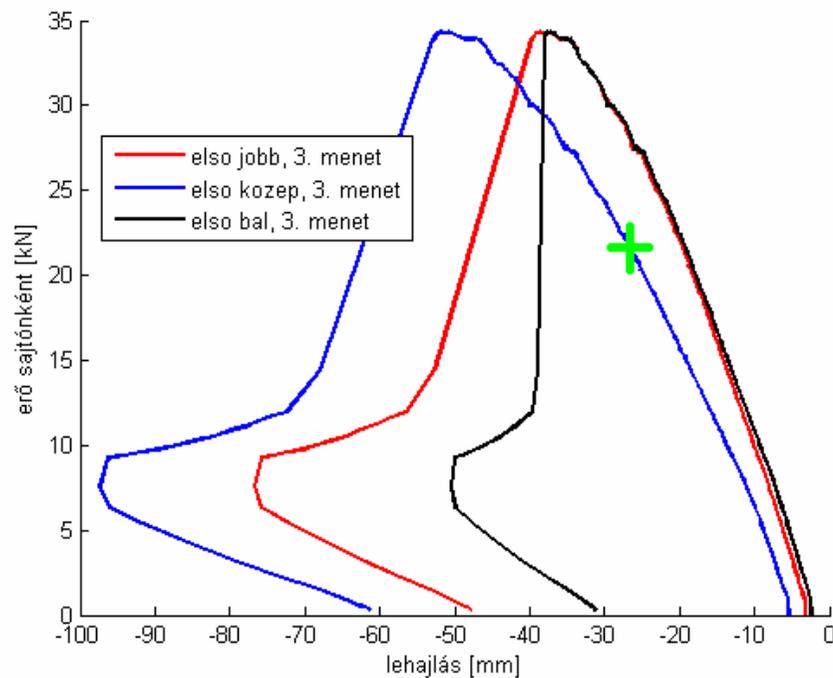
failure of the upper chord;
interaction of bending and
flexural buckling

Load: 47,4 kN/jack

ULS load: 31.65 kN/jack

LABORATORY TESTING – FE-MODEL

SLS level



ULS level	Axial force [kN]	
	FE	test
Upper chord	144,76	135,56
Lower chord	137,96	129,78
Brace	46,97	50,20



The model is applicable for design

LABORATORY TESTING – DESIGN METHOD

Compression chord member failure mode identified

- EC3 design rule modified based on strain measurement

$$\frac{N_{Ed}}{\chi_z \cdot A_{eff} \cdot f_{yb} / \gamma_{M1}} + \frac{\kappa_y \cdot M_{y,Ed}}{W_{eff,y} \cdot f_{yb} / \gamma_{M1}} + \frac{\kappa_z \cdot N_{Ed} \cdot 0.5 \cdot y_{sp}}{W_{eff,z} \cdot f_{yb} / \gamma_{M1}} \leq 1$$

Brace member failure modes identified

- EC3 design rule modified – calibrated – to ensure safety level

Joint failure mode identified

- EC3-based design formulae developed based on the existing design method of N-type RHS-RHS joints



AN EXAMPLE



FURTHER STEPS

Global FE surface model under development

Recently finished test series of cold-formed C sections with different end supports to study the truss members individually

DimTruss – a program to design these trusses – under development



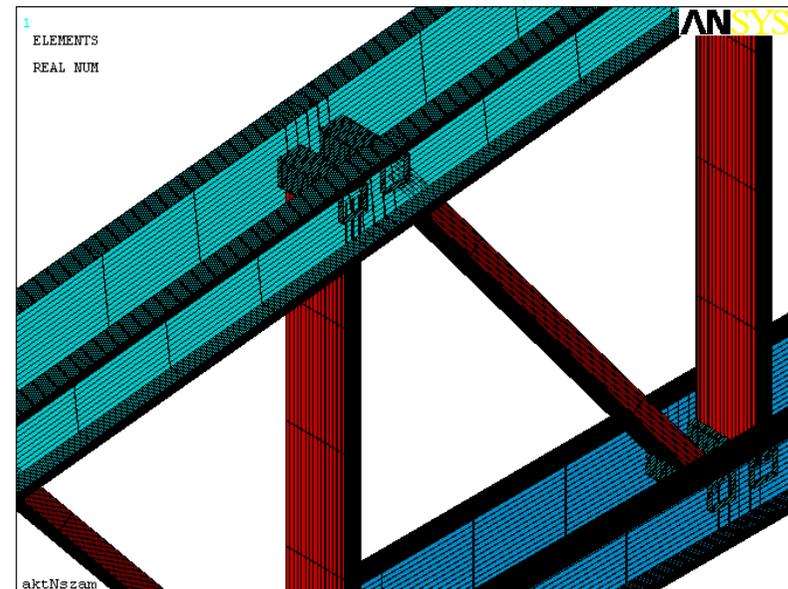
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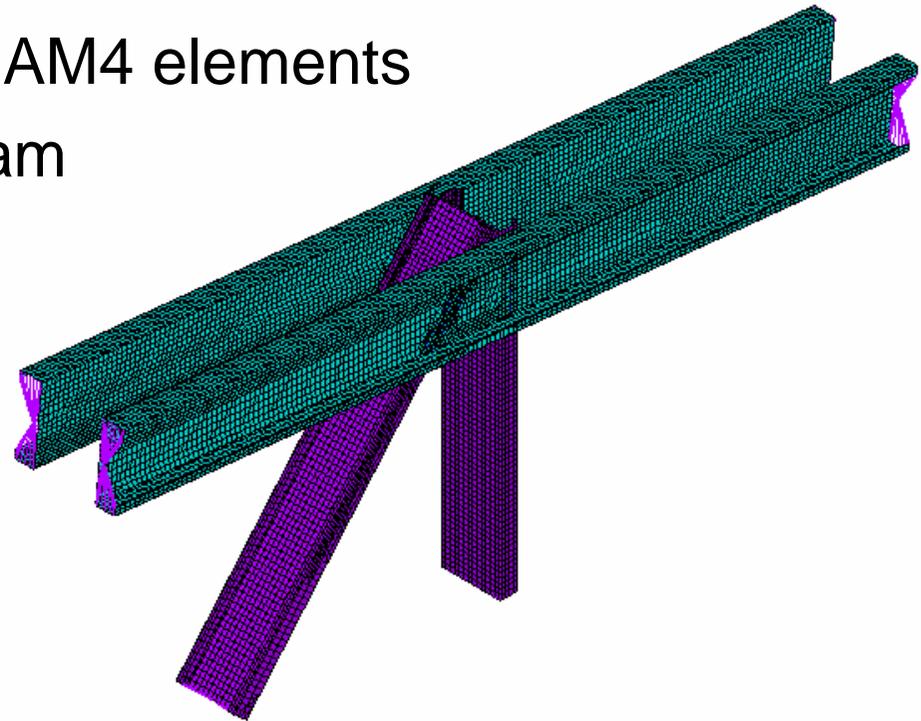
FE MODELLING – BEAM MODEL No. 2

- Global model – for development
 - Developed in ANSYS
 - 3-D beam model taking into account in-plane and out-of-plane eccentricities
 - 6 DOF's BEAM188 elements
 - Used to examine the joint area using submodeling technique



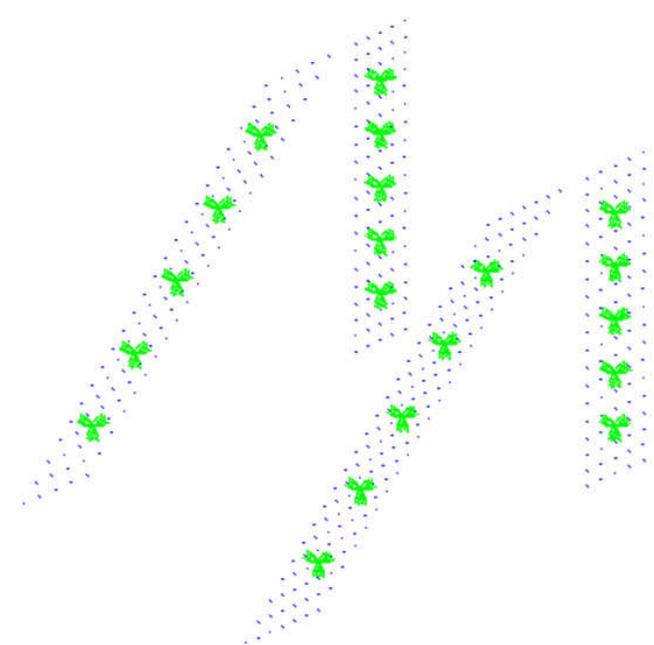
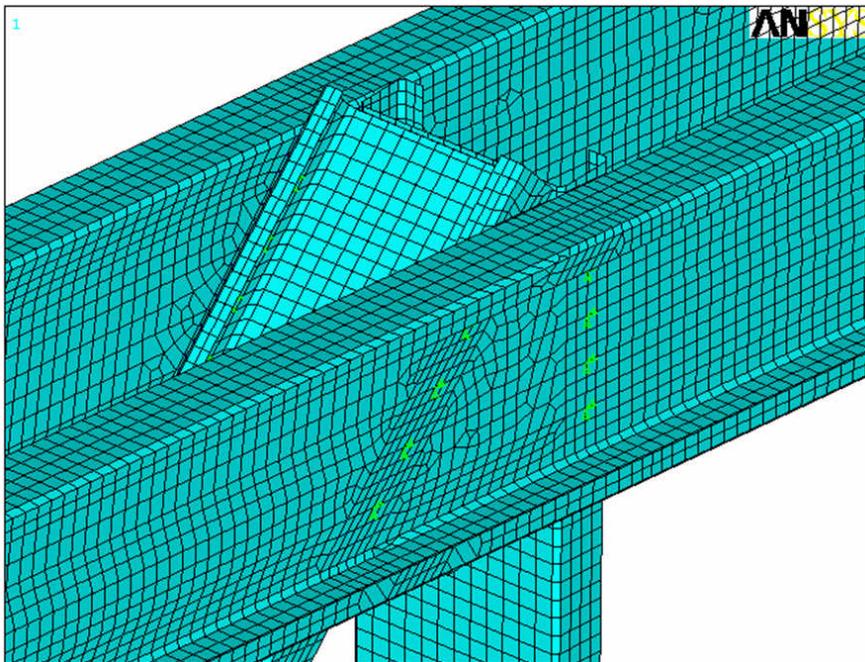
FE MODELLING – JOINT SURFACE MODEL

- Local model – for development
 - Developed in ANSYS
 - SHELL181, LINK10, BEAM4 elements
 - Kinematic load from beam model Nr. 2
 - ~100k DOF's
 - GMNIA



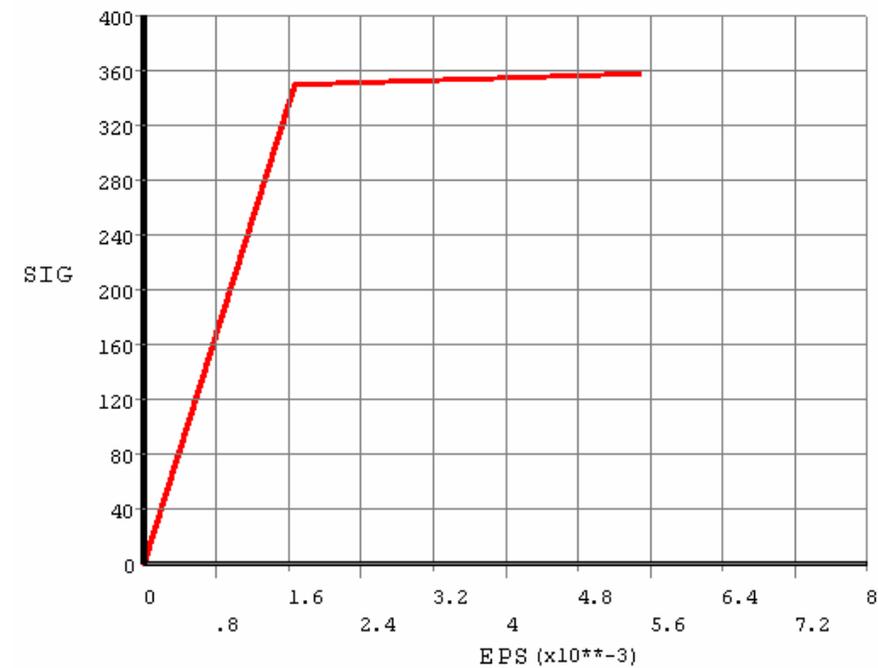
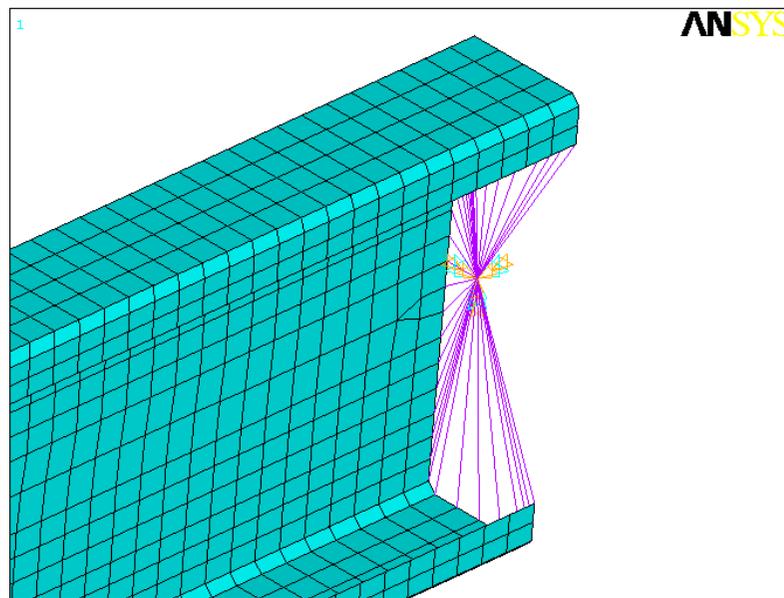
FE MODELLING – JOINT MODEL

- CONTACT AREA, BOLTS

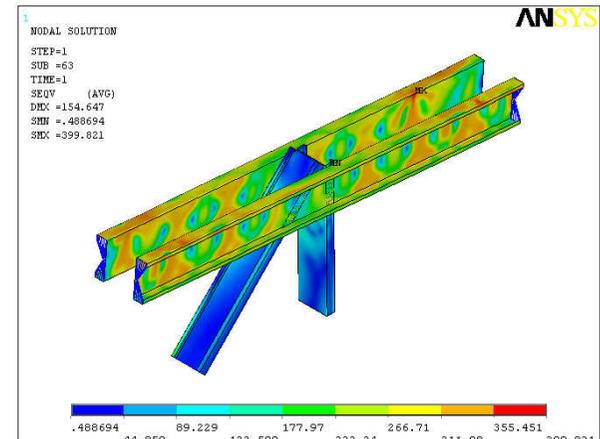
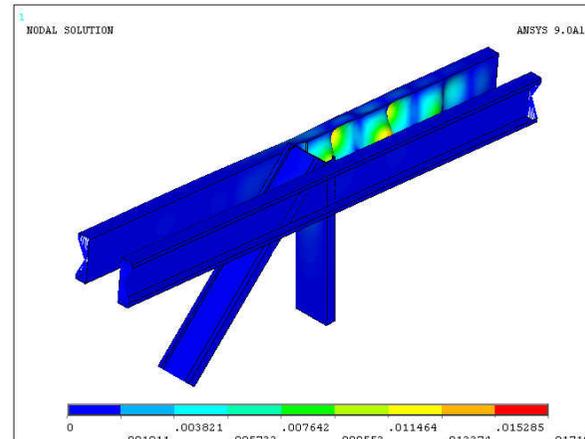
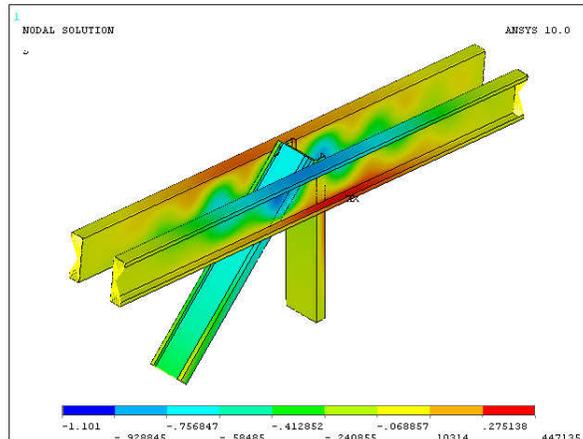
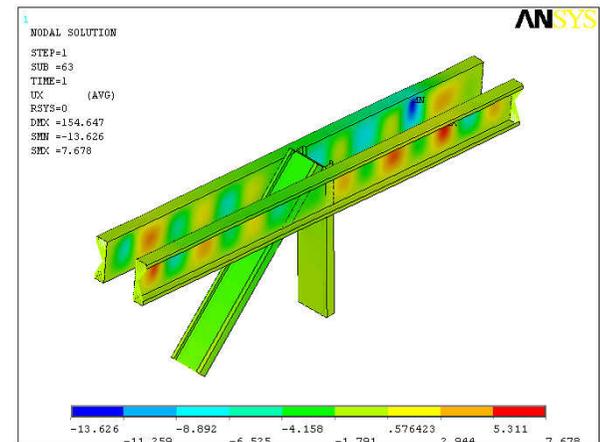
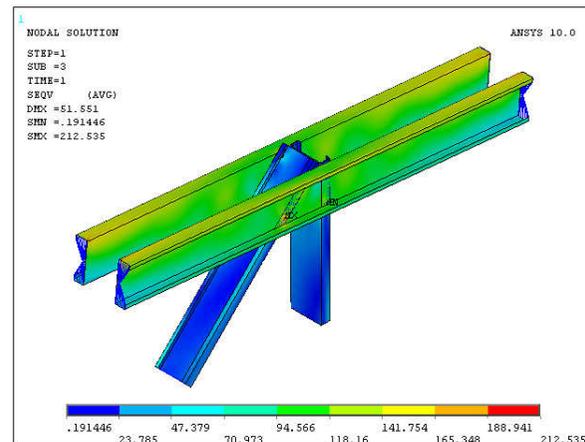
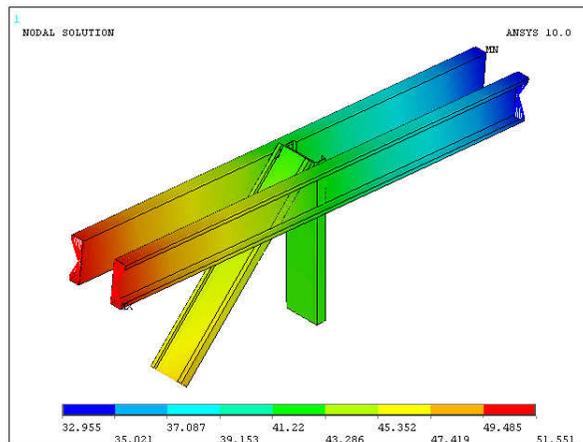


FE MODELLING – JOINT MODEL

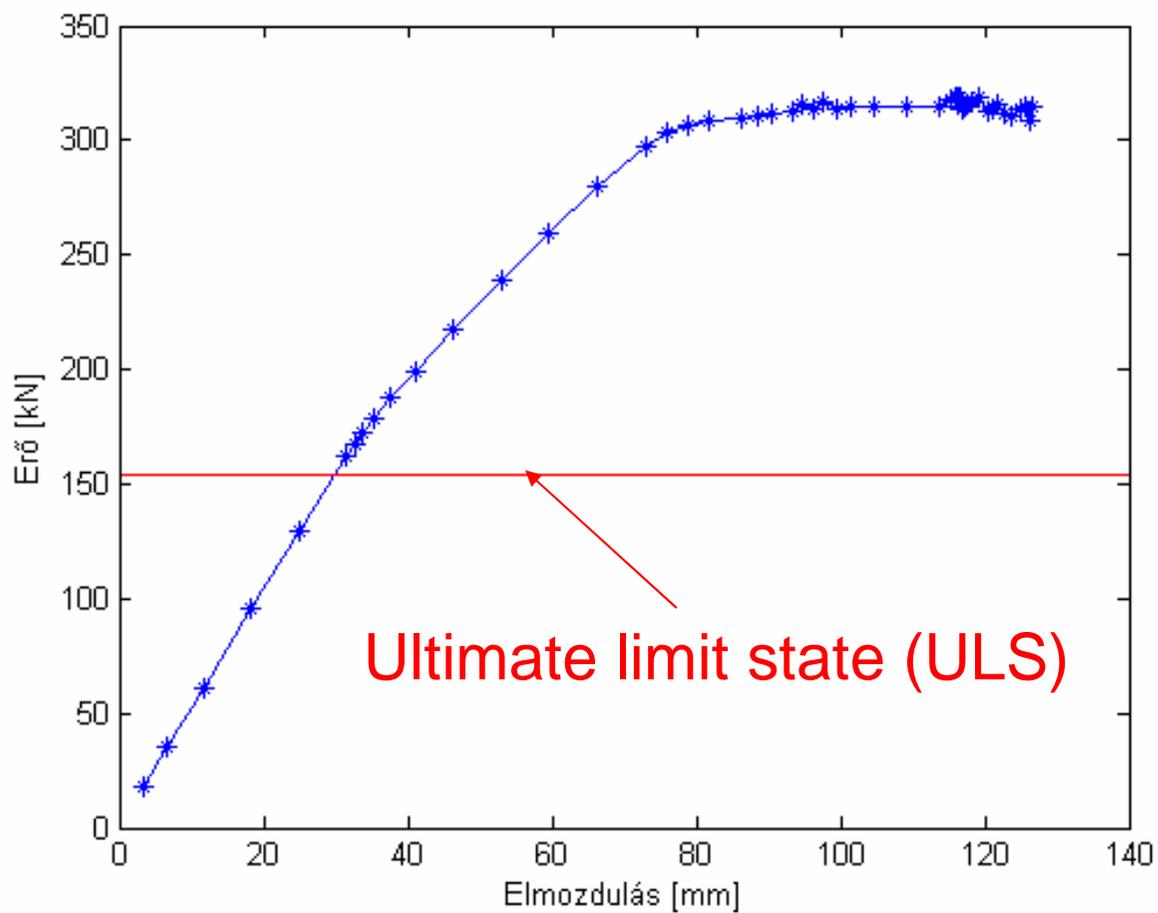
- LOADS, MATERIAL PROPERTIES



JOINT MODEL - RESULTS



JOINT MODEL - RESULTS



DESIGN METHOD – TENSION ELEMENTS

BRACE MEMBERS

- Tension and bending about the weak axis
- Plastic design resistance reduced

$$N_{pl,Rd} = 0,6 \cdot \frac{A_g \cdot f_{yb}}{\gamma_{M0}}$$

CHORD MEMBERS

- Tension and biaxial bending